## Errata: "A Model of Composite Propellant Combustion Including Surface Heterogeneity and Heat Generation"

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[AIAA J. 4, 1629–1637 (1966)]

RQUATIONS (8) and (15) should read

$$m_T = \rho_p \cdot r = M_T/S_0 = k_f/S_0 + k_{0x}/S_0 + k_{sr}(C)/S_0$$
 (8)

$$(\epsilon_i/D_i) = VP^{\delta}[(c_3/r) - (K_0D_i^n/P^m)]$$
(15)

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Quoted Ref. 17 should read:

<sup>17</sup> Pearson, G. S. and Sutton, D., "Ignition of composite propellant fuels by perchloric acid vapor," AIAA J. 4, 954–955 (1966).

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